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2021 MILITARY & AEROSPACE EMC GUIDE



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TABLE OF CONTENTS

EMC Equipment Manufacturers					
Summary of Military and Aerospace EMC Tests GHERY PETTIT Pettit EMC Consulting	7				
Overview of the DO-160 standard PATRICK ALBERSMAN Aerospace Pal, LLC - Founder	13				
Review of MIL-STD-461 CS118 Electrostatic Discharge STEVE FERGUSON Compliance Direction, LLC- principal consultant	19				
Selecting the Proper EMI Filter Circuit for Military and Defense Applications DAVE STANIS WEMS Electronics, ret.	22				
REFERENCE SECTION	28				
References (article links, directories, conferences, & Linkedin groups)	28				
Table of New Equipment	29				
Military Related Documents and Standards	31				
Aerospace Standards	32				
Index of Advertisers	33				

EMC EQUIPMENT MANUFACTURERS

Introduction

The following chart is a quick reference guide of test equipment and includes everything you'll need from the bare minimum required for key evaluation testing, probing, and troubleshooting, to setting up a full in-house precompliance or full compliance test lab for military and aerospace testing. The list includes amplifiers, antennas, current probes, ESD simulators, LISNs, near field probes, RF signal generators, spectrum analyzers, EMI receivers, and TEM cells. Equipment rental companies are also listed. The products listed can help you evaluate radiated and conducted emissions, radiated and conducted immunity and a host of other immunity tests, such as the new ESD test for MIL-STD-461G.



EMC Equipment Manufacturers					Type of Product/Service													
Manufacturer	Contact Information - URL			Chokes	Conducted Immunity	Current Probes	EMC Filters	EMC Testing	ESD Simulators	LISNs	Near Field Probes	Pre-Compliance Test	Radiated Immunity	Rental Companies	RF Signal Generators	Software	Spectrum Analyzers/EMI Receivers	TEM Cells
A.H. Systems	www.ahsystems.com	X	X			X						X				$\mid \mid \mid$		
Aaronia AG	www.aaronia.com	X	X									X					X	
Advanced lest Equipment Rentals	www.atecorp.com	Х	Х		Х	Х			Х	X	Х	Х	Х	Х	Х		X	X
ALTAIR	www.altair.com															X		
Anritsu	www.anritsu.com											χ			χ		X	
AR RF/Microwave Instrumentation	www.arworld.us	Х	Х		X			Х		X	Х	Х	Х		Х	X	Х	
Coilcraft	www.coilcraft.com			X														
Electro Rent	www.electrorent.com	Х			Х				χ	X		χ	χ	χ	Х		X	
EM Test	www.emtest.com/home.php				χ							χ						X
EMC Partner	www.emc-partner.com				χ				χ									
Empower RF Systems	www.empowerrf.com	χ											χ					
Exodus Advanced		v	v					v										
Communications	www.exoduscomm.com	^	^					^										
Fischer Custom Communications	www.fischercc.com					X				Х	Х	Х						
Gauss Instruments	www.gauss-instruments.com/en/																X	
Haefley-Hippotronics	www.haefely-hipotronics.com				Х				χ									
Instrument Rental Labs	www.testequip.com	χ			χ				χ	χ		χ	χ	χ	χ		X	
Instruments For Industry (IFI)	www.ifi.com	Х			Х								Х					
ITG Electronics	www.itg-electronics.com						X											
Keysight Technologies	www.keysight.com/main/home.jspx?cc=US&lc=eng									χ	χ	χ			χ	χ	X	
Microlease	www.microlease.com/us/home	χ			χ				χ	χ		χ	χ	χ	χ		X	
Milmega	www.milmega.co.uk	Х			Х								χ					
Narda/PMM	www.narda-sts.it/narda/default_en.asp	Х	Х		χ					X		χ	χ				X	
Noiseken	www.noiseken.com				χ				χ			χ						
Ophir RF	www.ophirrf.com	Х			Х													
Pearson Electronics	www.pearsonelectronics.com					X												
PPM Test	www.ppmtest.com		Х									χ	X			X	X	
R&B Laboratory	www.rblaboratory.com							X										
Rigol Technologies	www.rigolna.com					X					X	χ			χ	X	X	
Rohde & Schwarz	www.rohde-schwarz.com/us/home_48230.html	Х	Х		X	X				X	X	X	X		Х	X	X	
Siglent Technologies	www.siglentamerica.com										X	X			Х	X	X	
Signal Hound	www.signalhound.com										X	X			Х	X	X	
TekBox Technologies	www.tekbox.net	Х								X	X	χ				X		X
Tektronix	www.tek.com										X	X				X	X	
Teseq	www.teseq.com/en/index.php	Х			X	X			Х			X	X					X
Test Equity	www.testequity.com/leasing/	X			X				Х	X		Х	X	X	Х		X	
Thermo Keytek	www.thermofisher.com/us/en/home.html				X				Х								\square	
Thurlby Thandar (AIM-TTi)	www.aimtti.us											X			Х		X	\square
Toyotech (Toyo)	www.toyotechus.com/emc-electromagnetic-compatibility/	Х	X							X		Х	X				X	
TPI	www.rt-consultant.com											X			X		$\mid = \mid$	
Transient Specialists	www.transientspecialists.com				X								X				$\mid \mid \mid$	X
TRSRenTelCo	www.trs-rentelco.com/SubCategory/EMC_Test_Equipment.aspx	X	X		X					X		X	X	X	Х	\square	X	$\mid = \mid$
Vectawave Technology	www.vectawave.com	X															$\mid = \mid$	\mid
Windtreak Technologies	www.windtreaktech.com											X			Х			

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2



SUMMARY OF MILITARY AND AEROSPACE EMC TESTS

Ghery Pettit

Pettit EMC Consulting Ghery@PettitEMCConsulting.com

Introduction

Military and aerospace EMC tests cover a wide range of products. While the standards, including limits and test methods may differ, all EMC test standards have a few things in common. The most basic are the limits for emissions and the types and levels of susceptibility testing.

Emissions tests (and their associated limits) are put in place for military and aerospace equipment primarily to protect other systems from interference. These other systems may or may not include radio equipment. Examples abound showing the effect of inadequate EMC design. The Interference Technology 2016 Military EMC Guide (Reference 1) provides 3 such examples on page 11.



SUMMARY OF MILITARY AND AEROSPACE EMC TESTS

While many military and aerospace EMC issues may be addressed by operational changes, testing is still required to find weaknesses.

Military and aerospace EMC testing is performed at the system and subsystem levels. MIL-STD-464C provides requirements at the system or platform level. The latest version, MIL-STD-461G, provides requirements at the equipment or subsystem level. *Reference 1* provides details on both of the standards, but this article will highlight some key tests, particularly as they relate to MIL-STD-461G.

Ratio	Description
CE101	Conducted Emissions, Audio Frequency Currents, Power Leads
CE102	Conducted Emissions, Radio Frequency Potentials, Power Leads
CE106	Conducted Emissions, Antenna Port
CS101	Conducted Susceptibility, Power Leads
CS103	Conducted Susceptibility, Antenna Port, Intermodulation
CS104	Conducted Susceptibility, Antenna Port, Rejection of Undesired Signals
CS105	Conducted Susceptibility, Antenna Port, Cross-Modulation
CS109	Conducted Susceptibility, Structure Current
CS114	Conducted Susceptibility, Bulk Cable Injection
CS115	Conducted Susceptibility, Bulk Cable Injection, Impulse Excitation
CS116	Conducted Susceptibility, Damped Sinusoidal Transients, Cables and Power Leads
CS117	Conducted Susceptibility, Lightning Induced Transients, Cables and Power Leads
CS118	Conducted Susceptibility, Personnel Borne Electrostatic Discharge
RE101	Radiated Emissions, Magnetic Field
RE102	Radiated Emissions, Electric Field
RE103	Radiated Emissions, Antenna Spurious and Harmonic Outputs
RS101	Radiated Susceptibility, Magnetic Field
RS103	Radiated Susceptibility, Electric Field
RS105	Radiated Susceptibility, Transient Electromagnetic Field

MIL-STD-461G divides test requirements into 4 basic types. Conducted Emissions (CE), Conducted Susceptibility (CS), Radiated Emissions (RE) and Radiated Susceptibility (RS). There are a number of tests in each category and *Table I*, taken from MIL-STD-461G Table IV, shows these test methods.

A brief description of each these tests will be provided below. These are summarized from a more detailed introduction to MIL-STD-461G, which is found in the *References 1, 2*, and 3. Keep in mind that a complete copy of MIL-STD-461G is 280 pages, so any information here is brief and the standard must be read and understood. A copy of MIL-STD-461G may be obtained free. See *Reference 4*.

CE101 Conducted Emissions, Audio Frequency Currents, Power Leads. CE101 is applicable from 30 Hz to 10 kHz for leads that obtain power from sources that are not part of the EUT. There is no requirement on output leads from power sources. Emission levels are determined by measuring the current present on each power lead. There is different intent behind this test based on the usage of equipment and the military service involved. The specific limits are based on application, input voltage, frequency, power and current.

CE102 Conducted Emissions, Radio Frequency Potentials, Power Leads. CE102 is applicable from 10 kHz to 10 MHz for leads that obtain power from sources that are not part of the EUT. There is no requirement on output leads from power sources. The lower frequency portion is to ensure EUT does not corrupt the power quality (allowable voltage distortion) on platform power buses. Voltage distortion is the basis for power quality so CE102 limit is in terms of voltage. The emission levels are determined by measuring voltage present at the output port of the LISN. Unlike CE101, CE102 limits are based on voltage. The basic limit is relaxed for increasing source voltages, but independent of current. Failure to meet the CE102 limits can often be traced to switching regulators and their harmonics.

CE106 Conducted Emissions, Antenna Port. CE106 is applicable from as low as 10 kHz to as high as 40 GHz (depending on the operating frequency) for antenna terminals of transmitters, receivers, and amplifiers and is designed to protect receivers on and off the platform from being degraded by antenna radiation from the EUT. CE106 is not applicable for permanently mounted antennas.

CS101 Conducted Susceptibility, Power Leads. CS101 is applicable from 30 Hz to 150 kHz for equipment and subsystem AC and DC power input leads. For DC powered equipment, CS101 is required over the entire 30 Hz to 150 kHz range. For AC powered equipment, CS101 is only required from the second harmonic of the equipment power frequency (120 Hz for 60 Hz equipment) to

Table 1: MIL-STD-461G Emission and Susceptibility Requirements

INTERFERENCE TECHNOLOGY

150 kHz. In general, CS101 is not required for AC powered equipment when the current draw is greater than 30 amps per phase. The exception is when the equipment operates at 150 kHz or less and has an operating sensitivity of 1 μ V or better. The intent is to ensure that performance is not degraded from ripple voltages on power source waveforms.

CS103, CS104 and CS105 Conducted Susceptibility, Antenna Port, Intermodulation, Rejection of Undesired Signals and Cross-Modulation. This series of receiver front-end tests include test methods for Intermodulation (CS103), Rejection of Undesired Signals (CS104) and Cross Modulation (CS105). They were designed for traditional tunable super-heterodyne type radio receivers. Due to the wide diversity of radio frequency subsystem designs being developed, the applicability of this type of requirement and appropriate limits need to be determined for each procurement. Also, requirements need to be specified that are consistent with the signal processing characteristics of the subsystem and the particular test procedures to be used to verify the requirement.

CS109 Conducted Susceptibility, Structure Current. CS109 is a highly specialized test applicable from 60 Hz to 100 kHz for very sensitive Navy shipboard equipment (1 μ V or better) such as tuned receivers operating over the frequency range of the test. Handheld equipment is exempt from CS109. The intent is to ensure that equipment does not respond to magnetic fields caused by currents flowing in platform structure. The limit is derived from operational problems due to current conducted on equipment cabinets and laboratory measurements of response characteristics of selected receivers.

CS114 Conducted Susceptibility, Bulk Cable Injection. CS114 is applicable from 10 kHz to 200 MHz for all electrical cables interfacing with the EUT enclosures.

CS115 Conducted Susceptibility, Bulk Cable Injection, Impulse Excitation. CS115 is applicable to all electrical cables interfacing with EUT enclosures. The primary concern is to protect equipment from fast rise and fall time transients that may be present due to platform switching operations and external transient environments such as lightning and electromagnetic pulse.

CS116 Conducted Susceptibility, Damped Sinusoidal Transients, Cables and Power Leads. CS116 is applicable to electrical cables interfacing with each EUT enclosure and also on each power lead. The concept is to simulate electrical current and voltage waveforms occurring in platforms from excitation of natural resonances with a control damped sine waveform.

CS117 Conducted Susceptibility, Lightning Induced Transients, Cables and Power Leads. CS117 is one of two new test methods added to MIL-STD-461G. CS117 is applicable to safety-critical equipment interfacing cables and also on each power lead. Applicability for surface ship equipment is limited to equipment located above deck or which includes interconnecting cables, which are routed above deck. The concept is to address the equipment-level indirect effects of lightning as outlined in MIL-STD-464 and it is not intended to address direct effects or nearby lightning strikes.

CS118 Conducted Susceptibility, Personnel Borne Electrostatic Discharge. CS118 is applicable to electrical, electronic, and electromechanical subsystems and equipment that have a man-machine interface. It should be noted that CS118 is not applicable to ordnance items. The concept is to simulate ESD caused by human contact and test points are chosen based on most likely human contact locations. Multiple test locations are based on points and surfaces which are easily accessible to operators during normal operations. Typical test points would be keyboard areas, switches, knobs, indicators, and connector shells as well as on each surface of the EUT.

RE101 Radiated Emissions, Magnetic Field. RE101 is applicable from 30 Hz to 100 kHz and is used to identify radiated emissions from equipment and subsystem enclosures, including electrical cable interfaces. RE101 is a specialized requirement, intended to control magnetic fields for applications where equipment is present in the installation, which is potentially sensitive to magnetic induction at lower frequencies.

RE102 Radiated Emissions, Electric Field. RE102 is applicable from 10 kHz to 18 GHz and is used to identify radiated emissions from the EUT and associated cables. It is intended to protect sensitive receivers from interference coupled through the antennas associated with the receiver.

RE103 Radiated Emissions, Antenna Spurious and Harmonic Outputs. RE103 may be used as an alternative for CE106 when testing transmitters with their intended antennas. CE106 should be used whenever possible. However, for systems using active antenna or when the antenna is not removable or the transmit power is too high, RE103 should be invoked. RE103 is applicable and essentially identical to CE106 for transmitters in the transmit mode in terms of frequency ranges and amplitude limits. The frequency range of test is based on the EUT operating frequency.

RS101 Radiated Susceptibility, Magnetic Field RS101 is a specialized test applicable from 30 Hz to 100 kHz for Army and Navy ground equipment having a minesweeping or mine detection capability, for Navy ships and submarines, that have an operating frequency of 100 kHz or less and an operating sensitivity of 1 μ V or better (such as 0.5 μ V), for Navy aircraft equipment installed on ASW capable aircraft, and external equipment on aircraft that are capable of being launched by electromagnetic launch

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systems. The requirement is not applicable for electromagnetic coupling via antennas. RS101 is intended to ensure that performance of equipment susceptible to low frequency magnetic fields is not degraded.

RS103 Radiated Susceptibility, Electric Field. RS103 is applicable from 2 MHz to 18 GHz in general, but the upper frequency can be as high as 40 GHz if specified by the procuring agency. It is applicable to both the EUT enclosures and EUT associated cabling. The primary concern is to ensure that equipment will operate without degradation in the presence of electromagnetic fields generated by antenna transmissions both onboard and external to the platform. The limits are platform dependent and are based on levels expected to be encountered during the service life of the equipment. It should be noted that RS103 may not necessarily be the worst case environment to which the equipment may be exposed.

RS105 Radiated Susceptibility, Transient Electromagnetic Field. RS105 is intended to demonstrate the ability of the EUT to withstand the fast rise time, free-field transient environment of EMP. RS105 applies for equipment enclosures which are directly exposed to the incident field outside of the platform structure or for equipment inside poorly shielded or unshielded platforms and the electrical interface cabling should be protected in shielded conduit.

Not all tests are required for each type of device or intended use environment. MIL-STD-461G provides a matrix in Table V showing how these tests are used based on the intended use of the device.

Eavipment and							Ту	pe c	of Pr	rodu	ict/	Serv	ice						
Subsystems Installed In, On, or Launched From the Following Platforms or Installations	CE101	CE102	CE106	CS101	CS103	CS104	CS105	CS109	CS114	CS115	CS116	CS117	CS118	REI01	RE102	RE103	RS101	RS103	RS105
Surface Ships	A	A	L	A	S	L	S	L	A	S	A	L	S	A	A	L	L	A	L
Submarines	A	A	L	Α	S	L	S	L	A	S	L	S	S	A	A	L	L	A	L
Aircraft, Army, Including Flight Line	A	A	L	A	s	S	s		A	A	A	L	A	A	A	L	A	A	L
Aircraft, Navy	L	A	L	A	S	S	S		A	A	A	L	A	L	A	L	L	A	L
Aircraft, Air Force		A	L	A	S	S	S		A	A	A	L	A		A	L		A	
Space Systems, Including Launch Vehicles		A	L	A	s	s	s		A	A	A	L	A		A	L		A	
Ground Army		A	L	A	S	S	S		A	A	A	S	A		A	L	L	A	
Ground Navy		A	L	A	S	S	S		A	A	A	S	A		A	L	L	A	L
Ground, Air Force		A	L	A	S	S	S		A	A	A		A		A	L		A	

Legend:

A: Applicable (in green)

L: Limited as specified in the individual sections of this standard. (in yellow) S: Procuring activity must specify in procurement documentation. (in red)

Table 2: MIL-STD-461G Requirement matrix

Again, the reader is referred to *References 1* through 3 for more details, or to MIL-STD-461G for the details of the standard (*Reference 4*). This guide also provides a list of standards that apply to various military equipment.

A popular and common aerospace EMC requirement required by the FAA for commercial aircraft is RTCA/DO-160, Environmental Conditions and Test Procedures for Airborne Equipment. The latest version is RTCA/DO-160 G, published on December 8, 2010, with Change 1 published on December 16, 2015. DO-160 covers far more than just EMC issues, but the EMC subjects covered include input power conducted emissions and susceptibility, transients, drop-outs and hold-up; voltage spikes to determine whether equipment can withstand the effects of voltage spikes arriving at the equipment on its power leads, either AC or DC; audio frequency conducted susceptibility to determine whether the equipment will accept frequency components of a magnitude normally expected when the equipment is installed in the A/C; induced signal susceptibility to determine whether the equipment interconnect circuit configuration will accept a level of induced voltages caused by the installation environment; RF emissions and susceptibility; lightning susceptibility; and electrostatic discharge susceptibility.

This document can be purchased from RTCA on their website (*Reference 5*). A manufacturer producing products subject to the requirements in RTCA/DO-160 should obtain a copy and ensure they have a complete understanding of the content of the document and that any laboratory testing to it is properly accredited.

Examples of differences in test equipment between commercial and military standards.

There are differences in test equipment used compared with commercial EMC tests. Some examples are provided below.

Where 50 μ H LISNs are universally required for commercial EMC tests, there are specific cases for CE01 and CE02 tests where a 5 μ H LISN is called out. Limits for CE101 tests are provided in dB μ A. LISNs are only used for line impedance stabilization. The measurements are taken with current probes. Limits for CE102, on the other hand, are given in dB μ V and measurements are taken in much the same way as for commercial standards with the receiver connected to the RF output port of one of the LISNs and the other RF output port(s) terminated in 50 Ohms. It should be noted that MIL-STD-461G calls out a 20 dB pad on the output of the LISN to protect the receiver from transients. This is not a requirement in the commercial standards, but is worth considering when setting up a laboratory for commercial testing, as well.

Military EMC standards, such as MIL-STD-461G will require the use of different antennas for radiated emis

INTERFERENCE TECHNOLOGY

sions testing. Commercial equipment standards, such as CISPR 32 and ANSI C63.4, require the use of linearly polarized antennas and do not contain requirements for magnetic field testing.

MIL-STD-461G, RE101, requires the use of a 13.3 cm loop sensor, not required in the commercial standards. A receiver capable of tuning from 30 Hz to 100 kHz is needed.

MIL-STD-461G, RE102, requires testing of radiated emissions to as low as 10 kHz. From 10 kHz to 30 MHz a 104 cm (41 inch) rod antenna is used. This frequency range is not covered in CISPR 32 or the FCC Rules for radiated emissions. Thus, the antenna and receiver requirements are different. From 30 MHz to 200 MHz a biconical antenna is used, also commonly used in commercial testing. From 200 MHz to 1 GHz a double ridge horn antenna is called out in 461G. This is different than the tuned dipole or log periodic dipole array antennas used for commercial testing.

The test procedures are also different for radiated emissions testing, requiring different laboratory set-ups and test facility types. No turntable is needed for MIL-STD-461G, nor is an antenna mast capable of moving the antenna over a range of heights.

MIL-STD-461G, RS103, can require significantly higher field intensities for radiated susceptibility testing. Where

CISPR 35 requires 3 V/m from 80 MHz to 1 GHz and at a few discrete frequencies up to 5 GHz (with the option of testing a few discrete frequencies at up to 30 V/m), MIL-STD-461G requires testing from 20 V/m to as high as 200 V/m over the range of 2 MHz to 40 GHz for certain equipment. Additional test equipment (signal generators, amplifiers, antennas, etc.) is required over that needed for commercial testing.

Each test in MIL-STD-461G requires its own unique test equipment. Some may be useable for commercial testing, others may not. If testing to MIL-STD-461G, ensure that the equipment is proper for the tests being performed. A detailed understanding of the requirements in MIL-STD-461G is required to ensure that the proper equipment is being used and the laboratory is following the appropriate processes.

References

- 1. 2016 Military EMC Guide, Interference Technology
- 2. Ken Javor, MIL-STD-461G: The "Compleat" Review, Interference Technology, April 2016
- 3. Ken Javor, Why Is There AIR (in MIL-STD-461G)?, Interference Technology, April 2016
- 4. MIL-STD-461G, December 2015, Defense Acquisition System
- 5. RTCA/DO-160G, RTCA, December 2010.



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		1

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OVERVIEW OF THE DO-160 STANDARD

Patrick Albersman

Aerospace Pal, LLC - Founder patrick@aerospacepal.com

Introduction

In aerospace, there is one standard that always seems to be popping up, DO-160. Aircraft suppliers are often complying with aviation authorities' regulations by testing their product to DO-160.

DO-160 is a standard that was published by the industry group Radio Technical Commission for Aeronautics known as RTCA. RTCA is not a government regulation, however the FAA, EASA and others will often cited RTCA/DO-160 as a means of compliance for certification. In fact, it is the de facto standard in aerospace environmental testing.

DO-160 includes both environmental plus EMC, but in this article, we'll provide an overview of just the EMC-related sections. In later blogs, we'll go through each section, describing the tests in more detail, along with specific challenges for each.

So what is in RTCA/DO-160?



OVERVIEW OF THE DO-160 STANDARD

DO-160 ENVIRONMENTAL TESTING

The first thing to note is that DO-160 is test procedure. It is not a requirement. It is not a handbook. Sure, it gives some guidance of what testing is applicable but at its core is the need for companies to standardize testing categories, methods, and procedures. With this standardization, aerospace suppliers can produce products that more easily get certified on multiple aircraft.

For example, if Bombardier, Embraer and Boeing all have different testing needs then a supplier needs to understand all three testing requirements. In addition, the testing facilities that test to these standards will also have to read and comply with all the different procedures.

With the advent and adoption of RTCA/DO-160, airlines, suppliers, testing facilities and airlines all benefit from the standardization of testing.

- Airlines are not required to maintain test standards, methods, and procedures
- Suppliers can produce products that comply with multiple aircraft platforms
- Test Facilities become more efficient and familiar with one set of testing
- Airlines recognize test pedigrees from a common standard

So, what does DO-160 look like?

RTCA/DO-160 TOC

- Sect 1 Purpose and Applicability
- Sect 2 Definitions of Terms
- Sect 3 Conditions of Tests
- Sect 4 Temperature and Altitude
- Sect 5 Temperature Variation
- Sect 6 Humidity
- Sect 7 Operational Shocks and Crash Safety
- Sect 8 Vibration
- Sect 9 Explosion Proofness
- Sect 10 Waterproofness
- Sect 11 Fluids Susceptibility
- Sect 12 Sand and Dust
- Sect 13 Fungus Resistance
- Sect 14 Salt Spray
- Sect 15 Magnetic Effect
- Sect 16 Power Input
- Sect 17 Voltage Spike
- Sect 18 Audio Frequency Conducted Susceptibility (Power Inputs)
- Sect 19 Induced Signal Susceptibility

- Sect 20.0 Radio Frequency Susceptibility (Radiated and Conducted)
- Sect 21.0 Emission of Radio Frequency Energy
- Sect 22.0 Lightning Induced Transient Susceptibility
- Sect 23.0 Lightning Direct Effects
- Sect 24.0 Icing
- Sect 25.0 Electrostatic Discharge
- · Sect 26.0 Fire, Flammability
- Appendix A Environmental Test Identification
- Appendix B Membership
- Appendix C Change Coordinators

After that quick preview, let's take a look at each EMC related section.

RTCA/DO-160 EMC SECTIONS

Section 15.0 – Magnetic Effect

Technically Magnetic Effect is part of EMC, just barely. This test measures your equipment's effect on critical flight sensor like a compass. The goal is to determine where your product can be located, relative to these aircraft sensors. There is rarely an issue with the test results from Magnetic Effect.



DO-160 Figure 15-1 Test Installation and Procedure

Section 16.0 – Power Input

Power input is the longest section of the standard stretching almost 70 pages. The tests are run depending on your products power source (i.e. 28VDC, 115VAC, 270VDC.) The tests in the section range from normal, abnormal and emergency operating voltages to voltage surges. It also can measure to AC harmonics, <u>current</u> inrush and power factor. This section encompasses all things about your power input lines.

Categories for AC equipment are sorted by the expected frequency range. For example, category A(CF) – is for power sources that stay at the center frequency, 400Hz. A(NF) is for Narrow Frequency (360 to 650 Hz) and A(WF) is for Wide Frequency (360 to 800 Hz.)



DO-160 Figure 16-1 Generic Test Setup Example

DC categories are "A" for sources with DC supplied from transformer-rectifier units, "B" for sources with significant battery capacitance, "D" 270VDC equipment and "Z" sources without constant battery capacitance.

Section 17.0 – Voltage Spike

Voltage Spike determines whether your product can tolerate the voltage spikes arriving at the unit's power leads (AC or DC). The main adverse effects to be anticipated are permanent damage, component failure, insulation breakdown, susceptibility degradation, or changes in equipment performance.

Voltage Spike is separated in two categories, category "A" applying a 600V spike or category "B" applying a 200V spike or twice the line voltage (whichever is less).



DO-160 Figure 17-2 Voltage Spike Test Setup, DC or single-phase AC

Section 18.0 – Audio Frequency Conducted Susceptibility (Power Inputs)

Audio Frequency determines whether your unit will tolerate frequency components normally seen during operation of the aircraft. These frequency components are typically harmonics of the power source fundamental frequency.

The categories for Audio Frequency mirror that or power input. They include R(CF), R(NF) or R(WF) for AC power sources and R, B, or Z for DC sources. Cat K(CF), K(NF)

or K(WF) may also be required for AC systems with higher distortion levels.

The test applies distortion to the primary power through an audio transformer.



DO-160 Figure 18-1 Test Setup for Audio Frequency Conducted Susceptibility Test (For AC and DC Power Lines, Differential Mode)

Section 19.0 – Induced Signal Susceptibility

Inducted signal susceptibility includes five tests that determine the effect of interfering signals related to the power frequency harmonics, audio frequency signals, and electrical transients created by other systems. The test simulates noise generated on other interconnecting bundles that are routed in close proximity to your unit's wire harness on the aircraft.

The categories include B, A, Z and C which include increasing levels of susceptibility. The most severe test of 5 tests involves a switching relay chattering noise on to closely wrapped wires



DO-160 Figure 19-4 Audio Frequency Electric Field Susceptibility Test Setup

Section 20.0 – Radio Frequency Susceptibility (Radiated and Conducted)

Radio Frequency Susceptibility's purpose is to determine whether your product will operate within when the unit and its cable are exposed to a RF field. RF susceptibility is actually two tests; Radiated and Conducted Susceptibility, often referred to as RS and CS respectively. RF noises is applied to the EUT in continuous wave, square

2021 MILITARY & AEROSPACE EMC GUIDE

wave and pulsed modulation modes.

Both tests ensure the products uninterrupted operation when it is installed in the aircraft. The section's categories includes a designator for the CS level first and the RS level second (i.e. "YG".)

Radiated Susceptibility uses an anechoic chamber and an antenna to blast the product with RF whereas Conducted Susceptibility uses an injection clamp in induce the noise onto the EUT's I/O cable.



DO-160 Figure 20-2 Radiated Susceptibility Test Setup

Section 21.0 – Emission of Radio Frequency Energy Radiated and Conducted Emissions is your typical emissions testing for aerospace products. Just like section 20.0, this section is broken into two tests; radiated and conducted. The purpose of Emission of Radio Frequency Energy is to put an upper limit on the amount of RF energy your equipment can emit on the airplane. This will ensure proper integration and operation of the aircraft.

The categories include varying levels of acceptable emissions based on the location of the equipment. Consideration is also given by the aircraft manufacturer to the nature of the product being designed.

A typical setup for the conducted emissions portion is shown below.



DO-160 Figure 21-6 Typical Setup for Conducted RF Interference Test

Section 22.0 – Lightning Induced Transient Susceptibility

Lightning Induced Transient Susceptibility is intended to simulate lightning events striking the airplane and coupling onto your product's interconnecting cables.

This section, sometimes referred to as indirect lightning, is severe in nature, often driving hundreds or thousands of amps into a single pin of your unit's connector.

The section is broken into 3 separate parts; pin injection, induced cable bundle strikes and multiple burst testing. Pin injection looks for hard failures in your product after the strike where as cable bundle and multiple burst often requires your unit to operate while the event is occurring. Each test includes different waveforms that have durations as long as 500 μ s. Levels 1 – 5 increase in severity, topping out at 3200 volts and 5000 amps. This section is often the hardest to pass.



DO-160 Figure 22-17 Typical Cable Induction Test Setup

Section 23.0 – Lightning Direct Effects

Direct Effects, contrary to indirect lightning, tests products that may be directly struck by lightning. This is limited to products that will be mounted on the exterior aircraft. The section includes a high voltage strike test and a high current test. The equipment is often not expected to survive but must not cause any unsafe condition to the aircraft.



DO-160 Figure 23-10 – Typical Installation for Measurements of Injected Transients

Section 25.0 – Electrostatic Discharge

Electrostatic Discharge or ESD is a test to ensure your products reliance to static charges. This section test relates to airborne equipment which may be involved in static electricity discharges from human contact. It is applicable for all products that are accessible during normal operation or maintenance of the aircraft. Though no applicable to connector pins, aircraft manufacturers often require it.



DO-160 Figure 25-2 Simplified Diagram of the ESD Generator

SUMMARY

The EMC sections of DO-160 thoroughly tests your products ability to survive and operate in normal and abnormal conditions seen on an aircraft. Passing this 500-page standard is not easy. Understanding the test and how it is applied to your design is critical.

I have written extensively at www.aerospacepal.com including a DO-160 video to help users understand. Please also continue to read my blog on Interference Technology Magazine as I dive deeper into the sections listed above.





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REVIEW OF MIL-STD 461 CS118 – ELECTROSTATIC DISCHARGE

Steve Ferguson

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Introduction

MIL-STD-461G, released in December 2015, added a test for Personnel Borne Electrostatic Discharge (ESD). Prior to this release, this type of test for electronic systems and sub-systems was managed at the system level under MIL-STD-464C released in 2010. Both of these standards contain similar requirements. MIL-STD-461G provides the details on testing and MIL-STD-464C established the compliance requirements without detail on how to verify compliance.

Prior to 2010, several ESD programs were in use by DoD to provide control measures and in some cases test methods to verify tolerance to ESD events were included. A couple of more common standards have been used and are still active:

- MIL-STD-1686 provided a ESD control program for non-ordnance electronic devices with reference to HBM (Human Body Model), MM (Machine Model and CDM (Charged Device Model). The various models are divided in classes based on sensitivity and marked to provide guidance on control measure implementation. ANSI and IEC standards for HBM evaluation are referenced but newer versions are used than cited in the current MIL-STD-1686C.
- MIL-STD-331D is the current standard for ordnance ESD control and includes a different model than personnel borne ESD. The difference is associated with control measures required for ordnance handling. The standard refers to JOTP 062 for ordnance ESD testing and requirements.



REVIEW OF MIL-STD 461 CS118 – ELECTROSTATIC DISCHARGE

Background

From our high school years, physics training taught us about the atomic structure with electrons orbiting the nucleus and the net charge being related to the number of electrons relative to the number of protons. Materials with an unbound electron can release the electron to another atom with a small amount of energy. Contacting two materials can support the electron migration and separation of the materials may leave electrons trapped in another atom creating a net static charge between the two materials.

Depending on the forces present the static charge can reach very high voltage levels. If the voltage levels reach the point that the separating insulation allows the electron charge to recombine, a spark occurs bringing the two materials to a neutral charge. The static charge is created by many forces such as:

- Tribocharging where two bodies in contact are separated without allowing the electrons to return to their atoms. We encounter this from various sources such as walking across a carpet with each step supplementing our body charge. See *Figure 1* for a conceptual sketch. The charge will dissipate through the air or we contact an object that provides the conduction path with a transient discharge. Note that moisture molecules in the air provide for dissipating the charge through the air more easily, so in a humid environment the charge accumulation is typically reduced.
- Electrostatic induction where an object is placed near a charged object and the field of the charged object causes electron redistribution. This charged object is present with an open circuit provides a voltage without current flowing.
- Particles bombarding an object causing surface charging. Electrons are dislodged and moved in the direction of the particle movement creating the charge differential. This charge is often associated with moving objects or wind driven water referred to as precipitation static.



Figure 1: ESD Charge Accumulation Concept

We realize that ESD events occur when a breakdown potential of the insulation is reached. Keep in mind that air is an insulator as well as non-conducting materials. The breakdown potential of air is approximately 80V/mil of distance. The breakdown potential for most insulating material is approximately 200V/mil of thickness. So, based on these approximations, a 100 mil air gap or a 40 mil thick insulator should prevent a discharge with an 8kV charge. Damage to an insulator if subjected to a discharge can produce small "pin-holes" and the insulator becomes an air gap with significantly less insulating properties.

CS118 Calibration Verification

As with most tests in MIL-STD-461G, CS118 begins with the calibration verification or signal integrity check. *Figure 2* provides a general configuration for the calibration verification with two different checks to be accomplished. The ESD generator tip voltage is checked with an electrostatic voltmeter. The ESD generator is set for 2kV and the tip is placed at the measurement distance specified for the particular electrostatic voltmeter being used. Most electrostatic voltmeters have a guide or light convergence method to aid in placing the sensor at the correct distance. The voltmeter displays the measured voltage. The tip voltage tolerance is 10%. The tip voltage is repeated for each potential test voltage (4kV, 8kV, 15kV). If the tip voltage is not within tolerance correct the issue and redo the calibration verification.

The discharge current waveform is the second part of the calibration verification. *Figure 2* includes a configuration drawing for the discharge current check with the target schematic. The ESD generator is charged to the 8kV in the contact mode for the discharge current waveform check.



Figure 2: CS118 Calibration Verification Configurations

Set the oscilloscope to capture the waveform and place the ESD generator contact tip against the target plate. Trigger the discharge and review the captured waveform for compliance with the standard. The waveform shape is shown in *Figure 3* and detailed parameters are provided in *Table 1*. Note that the target design is the same as used for IEC 61000-4-2 testing standard.



Figure 3: Discharge Current Waveform

Displayed Voltage (KV)	First Peak Current, ±15% (A)	Rise time ^{1/} (ns)	Current I ₁ , ±30% (A) at t ₁ = 30 ns	Current I ₂ ±30% (A) at t ₂ = 60 ns				
±8	30	$0.6 \le t_r \le 1.0$	16	8				
1/Rise time is defined as the time from 10% to 90% of the peak value of the current waveform.								

Table 1: Discharge Current Waveform Parameters

CS118 Test

The test configuration for ESD testing uses the standard configuration without tailoring. This differs from the customized ESD configuration called out in the similar IEC standard, so if you are familiar with IEC testing, pay attention to this difference.

Connect the ESD generator ground strap to the EUT's chassis ground point. Only contact discharge at 8kV is required for conductive surfaces. Set the ESD generator to the test voltage and set for contact discharge. Apply 5-positive and 5-negative discharges while monitoring the EUT performance for indications of susceptibility.

Air discharge testing is applicable to test points where a contact discharge cannot be applied. Set the test voltage at 2kV for air discharge tests and apply 5-positive and 5-negative discharges to the test point. Note that discharges may not occur for each test point, but a residual charge could be present. Use a discharge conductor routed through a 1 M Ω resistor to ground to remove the charge between discharge applications. After testing each point at the 2kV level repeat testing at each of the higher test levels (4kV, 8kV, 15kV) to determine compliance.

The air discharge is accomplished by moving the ESD generator air tip toward the test point at a rate up to 0.3 meters/sec until discharge or contact to the test point oc-

curs. During test maintain a perpendicular orientation between the tip and the test point.

Test points should include locations likely to be contacted by the operator during normal use. The standard states that "test points to be considered shall include the following locations as applicable: any conductive or non-conductive points in the control or keyboard area and any other point of human contact such as switches, knobs, buttons, indicators LEDs, seams slots, grilles, connector shells and other accessible areas. As a minimum, each face shall be included." Recall that air serving as the insulator has a lower breakdown voltage that insulating material so be sure to include ventilation openings as test points.

Earlier in configuring the test, the ESD generator ground strap was to be connected to the EUT chassis. Nothing is mentioned about this connection for equipment with an ungrounded chassis such as portable or battery powered units. In these cases, the EUT generator ground strap would be connected to the test location ground reference plane. Also note that residual charge removal between discharges is very important to prevent over-testing from charge accumulation.

Summary

ESD events are common and often are not noticed where the event occurs with low voltage levels but still can produce defects in sensitive circuits. A good control program is necessary to minimize issues stemming from ESD. Factories where circuit exposure is normal incorporate many controls and continually verify that the control measures are followed.

Prevention of charge accumulation is a control measure that may be built into the product or installation. Maintaining conductivity between items provides the path for electrons to recombine and neutralize the charge. Providing a means to discharge with the unit allowing the personnel borne charge to be dissipated to a point not subject to damage.

Product testing allows us to have confidence that our devices have a reasonable ability to tolerate ESD and continue to perform as intended. This test addition to MIL-STD-461G provides a logical location for this evaluation and it provides the detailed test instructions to verify compliance.

SELECTING THE PROPER EMI FILTER CIRCUIT FOR MILITARY AND DEFENSE APPLICATIONS

Dave Stanis

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Introduction

Insertion loss, the term used to express a filter's ability to reduce or attenuate unwanted signals, has traditionally been measured in a 50 ohm source and 50 ohm load impedance condition, as standardized in MIL-STD-220.

In this matched 50 ohm impedance condition, various types of filter circuit configurations, single capacitor, "L's", "PI's", and "T's", will exhibit the same response for that given circuit regardless of the relationship between the input, output, and RF signal source.

MIL-STD-220 insertion loss tests are well defined, universal, and are excellent for monitoring filter manufacturing consistencies. However, the results can be misleading when it comes to selecting the proper filter circuit that must function in a complex impedance setting.



SELECTING THE PROPER EMI FILTER CIRCUIT FOR MILITARY AND DEFENSE APPLICATIONS

Introduction

Passive inductive and capacitive filters are impedance sensitive devices by nature and therefore source and load conditions must be taken into consideration when selecting a filter circuit.

This is particularly true, and becomes more pronounced, when you consider that most EMI line filters are not matched filter networks. That is to say the ideal design value of the individual components that make up the network have been modified, or intentionally mismatched, in order to accommodate operating line voltages, operating line currents, and reasonable packaging schemes.

In most cases the ideal inductor for a given response has been greatly reduced in value to accommodate the operating current and reduce the DCR; therefore the capacitors have to be increased in value to achieve the required insertion loss.

This intentional mismatch, which is widely practiced throughout the industry, only affects the very low frequencies by introducing ripple in the pass-band and has little, if any, negative effect in the reject band.

Circuit Configuration

EMI line filers are passive devices and their effect are bidirectional. They are all low-pass brute force networks, passing DC and power line frequencies with very low losses while attenuating the unwanted signals at higher frequencies.

They do not differentiate between EMI generated inside or outside the subsystem or system. They are equally effective in reducing EMI emissions as well as protecting a device from unwanted EMI entering via the power lines.

Each additional element improves the slope of the insertion loss curve. That is, the reject-band will be reached must faster with each section, or element, added. Increasing or decreasing the individual elements values does not change the slope of the curve but does affect the cutoff frequency.

More importantly, when the source and load impedance of the circuit changes, the slope of the insertion loss curve also changes. A "PI" circuit type filter, for example, is best suited when the source and load impedances are of similar values and relatively high. As these impedances become lower, the insertion loss for the "PI" filter also becomes lower. The reverse is true for "T" circuits.

If the circuit impedances varies with frequency, as most circuits do, then it is advantageous to use multiple ele-

ment filters such as a "PI" or "T" circuit. In the case of a "PI" circuit that exhibits maximum or load impedance is reduced the filter still has two active elements. For all practical purposes it becomes an "L" circuit. Additionally, the amount of filtering achievable is limited by the inductance (ESL) and resistance (ESR) in the capacitor and the parasitic capacitance in the inductors. The results are that the insertion loss curves "levels off" at approximately 80 to 90 dB.



Figure 1. Insertion Loss vs Frequency Curves

The following is a brief description of the most popular types of EMI Filter circuits and their application. It should be pointed out that these are only general guidelines due to the fact that most impedance conditions and EMI profiles are dynamic, complex, and change with frequency.

- Feedthrough Capacitor A single element shunt feedthrough capacitor has attenuation characteristics that increases at a rate of 20 dB per decade (10 dB at 10 kHz, 30 dB at 100 kHz). A feedthrough capacitor filter is usually the best choice for filtering lines that exhibit very high source and load impedances.
- L-Circuit Filter A two element network consisting of a series inductive component connected to a shunt feedthrough capacitor. This type of filter network has attenuation characteristics that increases at a rate of 40 dB per decade (20 dB at 100 kHz, 60 dB at 1MHz). An "L" circuit filter is best suited for filtering lines when the source and load impedances exhibit large differences. For most applications this type of network provides the greatest performance when the inductor is facing the lower of the two impedances.
- PI-Circuit Filter This is a three element filter consisting of two shunt feedthrough capacitors with a series inductive component connected between them. This three element filter has attenuation characteristics

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- that increases at a rate of 60 dB per decade (20 dB at 15 kHz, 80 dB at 150 kHz). A "PI" circuit filter is usually the best choice when high levels of attenuation are required and when the source and load impedances are of similar values and relatively high.
- T-Circuit Filter This also is a three element filter consisting of two inductive components with a single shunt feedthrough capacitors connected between them. Like the "PI" circuit filter, this device has attenuation characteristics that also increase at a rate of 60 dB per decade (20 dB at 15 kHz, 80 dB at 150 kHz). A "T" circuit filter is the best choice when high levels of attenuation are required and when the source and load impedances are of similar values and relatively low.
- Double Circuits Double "L's," double "PI's", and double "T's" consisting of four and five elements are best suited when extremely high levels of attenuation are required. Double "L's" have a theoretical attenuation of 80 dB per decade, while double "PI's" and double "T's" have a theoretical attenuation of 100 dB per decade. The source and load impedance conditions that apply to the single circuit devices apply to the double circuit filters.

The following table summarizes the various source and load impedance settings and the proper filter circuit for that condition.

Mismatching

As previously stated, most EMI line filters are intentionally mismatched for ease in manufacturing. A typical example of this industry wide practice is a cylindrical style filter.

The military specifications for this particular filer are:

Operating Voltage: 70 VDC

Operating Current: 5 ADC

Circuit Configuration: "PI"

DC Resistance: .015 ohms maximum

Case Diameter: .410 inches maximum

Full Load Insertion Loss per MIL-STD-220 (50 ohms):

150 kHz	300 kHz	1 MHz	10 MHz	100 MHz
16 dB	38 dB	75 dB	80 dB	80 dB

Based on a source and load impedance of 50 ohms, MIL-STD-220, a properly designed Butterworth filter (a filter network that has a maximum flat pass-band with average cutoff frequency to reject-band ratio), would produce the following element values in order to satisfy the minimum insertion loss requirements: C1 = .0769 μfd L2 = 385 μHy

C3 = .0769 µfd

The theoretical MIL-STD-220 insertion for a "PI" filter of these values is as indicated below:

50 kHz 3 dB	300 kHz 51 dB	1 MHz 83 dB	10 MHz >100 dB	100 MF >100 c
		Table 1		
Hi	gh lance	Circuit	Load Impeda	nce
Hi	gh	"C"	High	
Lo	w*	"L"	"L" High	
Hi	gh	"L"	Low	<u>.</u>
Hi	gh	"Pi"	High	
Lo	w	"T"	Low	
High	Low	"Double"	High / L	ow

The capacitance values for C1 and C3, .0769 μ fd, are acceptable for a 70 VDC rated filter and are easily manufactured. However, L2 must be 385 μ Hy in order to satisfy the insertion loss requirements.

In order to achieve 385μ Hy at 5 ADC, allow for core saturation (the change in incremental permeability of the core material with DC bias), and comply with the .015 DC resistance requirement, the diameter of the inductor would be in excess of 2.0 inches. This inductor would obviously not fit a case with an outside diameter of .410 inches.

By simply reducing the inductor to a realistic value and increasing the value of C1 and C3, we can achieve the required insertion loss in the reject-band with a design that can easily be manufactured. The typical values for this application would be:

C1 = .70 µfd
L2 = 5 µHy
C3 = .7 µfd

The theoretical MIL-STD-220 insertion for this modified filter is:

150 kHz	300 kHz	1 MHz	10 MHz	100 MHz
25 dB	50 dB	83 dB	>100 dB	>100 dB

As previously stated, this practice of intentionally mismatching the element values will introduce a substantial amount of ripple, as much as 10 to 20 dB, in the passband. However, at frequencies below 1 KHz, the response is normally flat to within \pm 1 dB.

Figure 2 depicts the MIL-STD-220 insertion loss characteristics for the ideal filter network and the modified design as compared to the specification requirements.



Figure 2. MIL-STD-220 insertion loss characteristics for ideal filter network and modified design compared to specification requirements.

MIL-STD-220 Insertion Loss Verses MIL-STD-461 EMI Testing

The majority of EMI filters are employed in order to cause system compliance to one of various military or commercial EMI/EMC specifications.

The most widely references military EMI/EMC specification is Military Specification MIL-STD-461 (462,463). This document specifies the allowable amount of conducted and radiated emissions that a subsystem or system can generate.

Conducted emissions is interference that is present, or 'conducted' on primary power lines (AC or DC) and/ or signal lines as detected by a current probe or other means. Radiated emissions is interference, both 'E" and "H" fields, that is being transmitted or radiated from the total system as detected by a receiving antenna.

In addition, MIL-STD-461 also delineates a series of tests that subject the device under test to various types of conducted and radiated interference to determine the survivability of the device when exposed to a harsh EMI environment. This series of tests is referred to as conducted and radiated susceptibility.

Conducted emission requirements and test methods are referred to as "CE". The numbers that follow refer to the applicable frequency range and whether it pertains to input power lines or signal lines. (i.e., CE03 establishes test methods and maximum allowable interference that can be present on AC and DC power lines over the frequency

range of 15 kHz to 50 MHz.) Similarly, "CS" stands for Conducted Susceptibility, "RE" for Radiated Emission, and "RS" for Radiated Susceptibility.

As previously stated, EMI filters being bidirectional devices not only help to reduce the amount of conducted emissions generated within, but also protect the system from unwanted interference entering via the power lines and signal lines.

To some degree EMI filers also help to reduce the radiated interference. This is due to the fact that the power lines and signal lines can act as 'transmitting antennas' if too much EMI is present. However, the majority of radiated problems are system configuration related (i.e., improper grounding, shielding, lack of EMI gaskets, the choice of materials in the case of "H" fields, etc.).



Figure 3. comparison of theoretical MIL-STD-220 50 ohm insertion loss of a "PI" filter and a "L" filter

The EMI profiles, and impedance, of any device is very complex and will change drastically over a given frequency range. It's this phenomenon that makes selecting an EMI filter based solely on 50 ohm insertion loss data difficult.

Figure 3 compares the theoretical MIL-STD-220 50 ohm insertion loss of a "PI" filter and a "L" filter comprised of the following components.

"PI" Circuit:

C1 = .70 μfd L2 = 5 μHy

C3 = .70 µfd

"L" Circuit:

C1 = .70 µfd

L2 = 5 µHy

2021 MILITARY & AEROSPACE EMC GUIDE

Looking at this comparison, and if size was not an issue, one would have a tendency to choose the "PI" circuit over the "L" circuit based on performance. At 1 MHz the "PI" circuit provides 80+ dB of insertion loss where the "L" circuit only provides 40+ dB.

However, MIL-STD-461 conducted emission tests are not performance under 50 ohm source and load conditions.

Figure 4 illustrates a typical MIL-STD-461 conducted emissions test configuration.



Figure 4. MIL-STD-461 Conducted Emissions Test Configuration

Not knowing the EMI source impedance (the device under test), we will assume ohms law. In this case 50 ohms. We don't know what the load impedance is, however, due to the 10 μ fd line stabilization capacitors (required by MIL-STD-461 as part of the test configuration), we can assume it is low compared to the source impedance. In this case, we will theorize 1 ohm.

In this more realistic setting, 50 ohm source and 1 ohm load, the "L" circuit performs almost as well as the "PI" circuit as illustrated in *Figure 5*. By slightly increasing the values of C1 and L2 in the "L" circuit, a response identical to the "PI" circuit can be achieved.





In the previous example we were only concerned with EMI emanating from the test sample. If we were also concerned about protecting against unwanted interference entering the device then a "T" circuit would be the filter of choice. In essence, by using a "T" circuit we have two "L" circuits with the inductor facing the lower impedance.

If the "T" circuit consisted of L1 facing the unit under test and, L3 facing the load with C2 in the middle, then for conduced emissions the "L" circuit is comprised of C2 and L3. For conducted susceptibility, if we assume the unit under test to be the lower of the two impedances, the "L" circuit is comprised of C2 and L1. In both instances the secondary inductor will provide some additional filtering. However, its contribution is relatively small compared to the other two components.

There are an infinite number of source and load impedance combinations for signal line applications where the 10 μ fd line stabilization capacitors are not required as part of the test configuration. For these situations the theoretical insertion loss can be calculated by varying RS and RL in the equations.

Although the circuits that we have been discussing only address common mode (interference which is present as a common potential between ground and all power lines) EMI, the same philosophies apply when selecting differential mode (interference which is present as a potential between individual power lines) EMI filtering elements commonly found in multicircuit filter assemblies, or "Black Box".

Conclusion

Selecting the proper EMI filter circuit is not a difficult task provided, that as a minimum, the following parameters are taken into consideration:

- The EMI source impedance
- The EMI load impedance
- The EMI propagation mode (common mode, differential mode or both)
- Conducted emission requirements
- Conducted susceptibility requirements

Other considerations that are not readily apparent are the effects caused by mismatching; performance at full load; and the inability to achieve the theoretical insertion loss due to the inductance (ESL) and resistance (ESR) in the capacitor, and the parasitic capacitance in the inductors.

For more information about EMI Filters and Filter Connectors, please contact:

Mike MacBrair

Vice President Sales and Marketing Cell: 310-956-0807 Office: 310-644-0251 ext. 110 Email: mmacbrair@wems.com



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REFERENCES (ARTICLE LINKS, DIRECTORIES, CONFERENCES, & LINKEDIN GROUPS)

LINKS TO LONGER ARTICLES

"MIL-STD-461G – The Compleat Review" https://interferencetechnology.com/mil-std-461gcompleat-review/

"Selecting the Proper EMI Filter Circuit For Military and Defense Applications"

https://interferencetechnology.com/selecting-properemi-filter-circuit-military-defense-applications/

"Why is there AIR (In MIL-STD-461G)?"

https://interferencetechnology.com/air-mil-std-461g/

"Overview of the DO-160 standard"

https://interferencetechnology.com/overview-of-the-do-160-standard/

"Review of MIL-STD-461 CS118 - Electrostatic Discharge"

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Test House Directory – 2016 Test and Design Guide http://learn.interferencetechnology.com/2016-emctest-and-design-guide/

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www.asdevents.com/shopcontent asp?type =aerospace_defence

Aviation Week Event Calendar: www.events.aviationweek.com/current/Public/Enter.aspx

Defense Conferences: www.defenseconference.com/

Global Edge (MSU): www.globaledge.msu.edu/industries/aerospace-anddefense/events/

ICMST Events (PDF): www.drdo.gov.in/drdo/egate/FICMST_Vol_12_No_03_ January_2017.pdf

IEEE AESS Events: www.ieee-aess.org/conferences/home

Jane's Events: www.janes.com/events

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RF and Microwave Community

TABLE OF NEW EQUIPMENT ALLOWED/REQUIRED IN MIL-STD-461G

Tony Keys EMC Analytical Services

Ken Javor EMC Compliance

The following table was compiled by Ken Javor, of EMC Compliance. The updated changes to MIL-STD-461G require some new equipment. One of these changes allows the use of time domain EMI receivers, which will help speed up the testing, due to their fast FFT-based signal acquisition. Following is a list of some specific changes and equipment requirements:

CS101 (Conducted Susceptibility, Power Leads) - There is now a requirement to measure induced AC power line ripple. This requires a new "power ripple detector", which is a specially designed isolation transformer that matches the power line to 50 ohms.

CS114 (Conducted Susceptibility, Bulk Cable Injection) - This injection probe test now requires the use of a current probe calibration fixture to validate the test level during pre-calibration.

CS117 (Conducted Susceptibility, Lightning Induced Transients, Cables and Power Leads) - This is a new test added to MIL-STD-461G and requires a lightning transient simulator.

CS118 (Conducted Susceptibility, Personnel Borne Electrostatic Discharge) - This is a new test added to MIL-STD-461G and requires a standard electrostatic discharge simulator.

RS103 (Radiated Susceptibility, Electric Field) - This test requires an E-field antenna that can go down to 2 MHz.



2021 MILITARY & AEROSPACE EMC GUIDE

Table of New Equipment Required for Latest Updates to MIL-STD-461G							
Requirement	Equipment Type	Vendor(s)	Websites				
General	Time Domain EMI Receivers*	Amplifier Research Gauss Instruments Keysight Rohde & Schwarz	http://www.arworld.us/html/dsp-receiver-multistar.asp http://www.gauss-instruments.com/en/products/tdemi http://www.keysight.com/en/pdx-x201870-pn-N9038A/mxe-emi-receiver-3-hz-to-44- ghz?cc=UG&lc=eng https://www.rohde-schwarz.com/us/products/test-measurement/emc-field-strength-test- solutions/emc-field-strength-test-solutions_105344.html				
CS101	Frequency Domain Ripple Monitoring Transducer [*] High-voltage Differential Probe, 100 MHz, 1k V(RMS) Digital Oscilloscopes (200 MHz - 4 GHZ, 5/10 GSa/s)	Pearson Electronics Rohde & Schwarz Rohde & Schwarz	http://www.pearsonelectronics.com/news/179 https://www.rohde-schwarz.com/us/product/rtzd01-productstartpage_63493-34629.html https://www.rohde-schwarz.com/us/product/rto-productstartpage_63493-10790.html or https://www.rohde-schwarz.com/vn/product/rte-productstartpage_63493-54848.html (with Option RTO-K17)				
CS114	Current Probe Calibration Fixture	ETS/Lindgren Fischer Custom Communications Pearson Electronics Solar Electronics	http://www.ets-lindgren.com/EMC (fixture not listed on web site but should be part of current probe/injection clamp line-up) http://www.fischercc.com/ViewProductGroup.aspx?productgroupid=141 http://www.pearsonelectronics.com/news/180 (fixture holds both injection clamp and current probe) http://www.solar-emc.com/RFI-EMI.html (scroll to bottom of page)				
CS117	Indirect Lightning Test Systems	HV Technologies Thermo Scientific Solar Electronics	http://www.hvtechnologies.com/TestsTrack/Lightning/tabid/408/Default http://www.thermoscientific.com/en/product/ecat-lightning-test-system-lts.html http://www.solar-emc.com/2654-2.html				
CS118	ESD Gun	EMC Partner EM Test Haefely Kikusui LISUN Group Noiseken Thermo Scientific TESEQ	https://www.emc-partner.com/products/immunity/esd/esd-generator http://www.emtest.com/products/productGroups/ESD_generators.php http://www.haefely-hipotronics.com/product/product-category/electrostatic-discharge-test- systems-esd/ http://www.kikusui.co.jp/en/product/detail.php?ldFamily=0020 http://www.lisungroup.com/product-id-318.html http://www.noiseken.com/modules/products/index.php?cat_id=1 http://www.thermoscientific.com/en/product/minizap-15-esd-simulator.html http://www.teseq.com/product-categories/esd-simulators.php				
RS103	1 – 18 GHz Electric Field Probe (most test facilities already have one)	Amplifier Research ETS/Lindgren NARDA	http://www.arworld.us/html/field-analyzers-field-monitoring.asp http://www.ets-lindgren.com/EMCProbes http://www.narda-sts.us/products_highfreq_bband.php				

* Specified as acceptable for use, but not required.

MILITARY RELATED DOCUMENTS AND STANDARDS

The following references are not intended to be all inclusive, but rather a representation of available sources of additional information and point of contacts.

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MIL-HDBK-237D Electromagnetic Environmental Effects and Spectrum Certification Guidance for the Acquisition Process, 20 May 2005.

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MIL-STD-188-125-1 High-Altitude Electromagnetic Pulse

(HEMP) Protection for Ground-Based C41 Facilities Performing Critical, Time-Urgent Missions Part 1 Fixed Facilities, 17 Jul 1998.

MIL-STD-220C Test Method Standard Method of Insertion Loss Measurement, 14 May 2009.

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MIL-STD-1541A Electromagnetic Compatibility Requirements for Space Systems, 30 Dec 1987.

MIL-STD-1542B Electromagnetic Compatibility and Grounding Requirements for Space System Facilities,

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15 Nov 1991. MIL-STD-1605 Procedures for Conducting a Shipboard Electromagnetic Interference (EMI) Survey (Surface Ships), 08 Oct 2009.

MIL-STD-1686C Electrostatic Discharge Control Program for Protection of Electrical and Electronic Parts, Assemblies, and Equipment (Excluding Electrically Initiated Explosive Devices). 25 Oct 1995.

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http://www.aiaa.org/default.aspx

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DO-233, Portable Electronic Devices Carried on Board Aircraft

DO-235B, Assessment of Radio Frequency Interference Relevant to the GNSS L1 Frequency Band

DO-292, Assessment of Radio Frequency Interference Relevant to the GNSS L5/E5A Frequency Band

DO-294C, Guidance on Allowing Transmitting Portable Electronic Devices (T-PEDs) on Aircraft

DO-307, Aircraft Design and Certification for Portable Electronic Device (PED) Tolerance

DO-307A, Aircraft Design and Certification for Portable Electronic Device (PED) Tolerance

DO-357, User Guide: Supplement to DO-160G

DO-363, Guidance for the Development of Portable Electronic Devices (PED) Tolerance for Civil Aircraft DO-364, Minimum Aviation System Performance Standards (MASPS) for Aeronautical Information/Meteorological Data Link Services

DO-363, Guidance for the Development of Portable Electronic Devices (PED) Tolerance for Civil Aircraft

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SAE Standards

http://www.sae.org/

ARP 5583 – Guide to Certification of Aircraft in a High Intensity Radiation (HIRF) Environment http://standards.sae.org/arp5583/



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